Radiation Shielding Simulation For Interplanetary Manned Missions

S. Guatelli¹, B. Mascialino¹, P. Nieminen², M.G. Pia¹



¹INFN Genova, Italy ²ESA-ESTEC, The Netherlands





IPRD 06 10th Topical Seminar on Innovative Particle and Radiation Detectors 1 - 5 October, 2006 Siena, Italy



Susanna Guatelli



Context

- Planetary exploration has grown into a major player in the vision of space science organizations like ESA and NASA
- AURORA: European long term plan for the robotic and human exploration of the solar system
 - with Mars, the Moon and the asteroids as the most likely targets



Human missions to Mars

The effects of space radiation on astronauts are an important concern



This project In the framework of the AURORA programme of ESA



Quantitative evaluation of the physical effects of space radiation in interplanetary manned missions



First **quantitative** evaluation of the shielding properties of **conceptual designs** Comparison among different shielding options

Guidelines for future mission design and concrete engineering studies

Strategy

Risk assessment	 Reliable model to evaluate acceptable risk (risk is not measured, it is predicted by a model)
Sound physics results	 Distinguish physical and biological effects Validation of the physics modelling components Control systematics Quantitative mathematical analysis of results
Advanced technology	 Object Oriented technology Rigorous software process Software tools based on modern technology
Transparency	 Open source software tools Publicly distributed software
All the features above are relev	vant to a mission-critical long-range space science project

Uncertainties and systematics

Large uncertainties

- Radiation environment
- Mission trajectories and duration
- Characteristics of spacecrafts
- Biological effects of radiation exposure

Modelling hadronic interactions

- Intrinsic complexity of the underlying theory
- Scarce and imprecise experimental data
- Monte Carlo codes still evolving in this domain



Relative comparison of shielding configurations Instead of absolute dose calculations

Multiple hadronic models to evaluate **systematics**

Software Strategy

Software Development Process

- Iterative and incremental model
- Based on the Unified Process
- Specifically tailored to the project
- Mapped onto ISO 15504

Monte Carlo simulation

- Model radiation environment
- Model shielding configuration
 - Track particles in matter





Functionality (geometry, physics, interactivity)

> **OO** technology Rigorous **validation**

Physical Analysis

Evaluate energy deposited in the astronaut



Abstract Interfaces for Data Analysis

Independence from any specific analysis tools

Object Oriented Technology

- Openness to extensions and evolution
- Maintainability over a long time scale

Statistical Analysis

Compare shielding options quantitatively



Richest collection of **goodness-of-fit** tests for data comparison

AIDA compliant

Architecture of the Geant4 application



Component-based architecture

Each component has a specific responsibilityFlexibility, extensibility and maintainability

Geometry and materials



Dynamically configure the simulation with

- Habitat type
- Shielding geometry
- Material
- Astronaut's location

Primary particle generation

Conservative assumptions for risk assessment

CREME96

Radiation environment models ECSS standard

Galactic Cosmic Rays

- 1977 solar minimum
- Anti-correlation with solar activity

Solar Particle Events

- October 1989 event
- 99% worst case

Susanna Guatelli



Electromagnetic physics

Physics modelling

- Geant4 Low Energy Package for:
 - p, α , ions (model based on ICRU49 parameterisation)
 - photons and electrons (model based on the Livermore Libraries)
- Geant4 Standard Package for positrons and muons
- Low Energy Electromagnetic Package overall better accuracy
 - Demonstrated by validation studies

Hadronic Physics

- Elastic and inelastic hadronic scattering
- Alternative models of the inelastic scattering of protons, neutrons, pions:

Hadronic inelastic process	Binary set	Bertini set
Low energy range	Binary Cascade + Precompound + Deexcitation	(up to 3.2 GeV)
deexcitation)	(up to 10. GeV)	
Intermediate energy range	Low Energy Pa	arameterised
(up to $E = 25 \text{ GeV}$)		
High energy range	Quark Gluon String Model	
(20. GeV < E < 100. GeV)	Quark Ordon (

- α inelastic scattering
 - Low Energy Parameterized model up to E = 100 MeV
 - Binary Ion model

Shielding studies



Inflatable Habitat

- Shielding materials
- Shielding thickness
- Comparison to rigid habitats





Planet surface habitats

- Shielding with lunar regolith
- Comparison to a rigid habitat



Engineering model by ALENIA SPAZIO

Target

Simplified geometry Retains the **essential characteristics** for a _{Susanna Guatelli} shielding characterization

Inflatable habitat

Multilayer structure

- External thermal protection blanket
 - Betacloth and mylar
- Meteoroid and debris protection
 - Nextel (bullet proof material) and open cell foam
- Structural layer
 - Kevlar
- Rebundant bladder
 - Polyethylene, polyacrylate, EVOH, kevlar, nomex



Systematics of hadronic modelling

Study of possible systematic effects related to hadronic physics modelling Comparison of the energy deposit in the phantom with different hadronic models





Different energy deposit profiles and secondary product patterns Kolmogorov Smirnov Test: p-value < 0.05

Physics models	Energy deposit in the phantom	
Electromagnetic	71.9 ± 0.2 MeV	~10 °
Hadronic-Bertini	128.0±0.4 MeV	of G
Hadronic-Binary	117.3±0.4	

~10 % difference **compatible** with **experimental validation** of Geant4 hadronic models

Shielding thickness

Impossible to shield galactic cosmic rays completely!





The shielding modulates the fraction of GCR reaching the phantom

Secondary particles are also generated in the shielding material

Effect of **5 cm** and **10 cm water shielding** + inflatable multi-layer

% GCR p reaching the phantom

	5 cm water	10 cm water
EM + Bertini	88.7 ± 0.2	83.3 ± 0.2
EM + Binary	88.0 ± 0.2	82.3 ± 0.2

Energy deposit in the phantom

	5 cm water	10 cm water
EM + Bertini	130.2 ± 0.5	128.0 ± 0.5
EM + Binary	119.3 ± 0.4	117.3 ± 0.4

Doubling the shielding thickness results in:

- Few % difference in the stopped GCR protons
- ~2 % difference in the total energy deposit

Shielding material for an inflatable habitat

0.07

0.06

0.05

0.04

0.03

0.02

0.01

Polyethylene

Water

Energy (MeV)



Water and polyethylene: equivalent shielding effect



Pions Alpha Other e- Gamma e+ N

nu

Inflatable vs. Rigid Habitat

Reference: rigid structures as in the ISS (2 - 4 cm Al)



Kolmogorov-Smirnov test

Multi-layer + 10 cm water equivalent to 4 cm Al
Multi-layer + 5 cm water equivalent to 2.15 cm Al

Shielding	Energy deposited in phantom (MeV)		
material	EM	Bertini	Binary
ML + 5 cm water	73.5 ± 0.3	130.2 ± 0.5	119.3 ± 0.4
ML + 10 cm water	71.9 ± 0.3	128.0 ± 0.5	117.3 ± 0.5
4 cm Al	72.9 ± 0.3	127.5 ± 0.5	117.0 ± 0.4
2.15 cm Al	73.9 ± 0.3	130.5 ± 0.5	119.3 ± 0.5

An inflatable habitat exhibits a shielding capability equivalent to a conventional rigid one

Susanna

Secondary products in inflatable and rigid habitats

Average no. of secondary particles reaching the phantom





Anderson-Darling test Cramer-von Mises test χ^2 test



No significant effects on the energy deposit in the phantom

Susanna Guatelli

Effects of other cosmic ray components

Similar studies of GCR α , C, O, Si and Fe ion components

Energy deposited in phantom (MeV) by GCR α Weighted according to the relative flux w.r.t. p

	Multi-layer + 10 cm water
EM	27.2 ± 0.1
EM + Bertini	32.0 ± 0.1
EM + Binary	31.7 ± 0.1

Dominant contribution derived from electromagnetic interactions

First indication

The hadronic models for α particles are under development and under validation

Average energy deposit of GCR ions Weighted according to the relative flux w.r.t. p



Main contribution from p and α

Radiation protection from solar particle events

A shelter protects the crew from the harmful effects of solar particle events



Shelter: limited zone enclosed by an additional shielding layer

Alarm of a solar particle event



The astronauts recover in the shelter

Solar particle events

Fraction of solar event protons reaching the phantom

Shelter thickness (cm)	%
0	(1.03 ± 0.05) 10 ⁻¹
20	(1.6 ± 0.2) 10 ⁻²
50	(3.4 ± 0.8) 10 ⁻³
75	(2.0 ± 0.6) 10 ⁻³





An additional shelter layer is effective at stopping a significant fraction of solar particle events

Radiation shielding on planetary surface

Moon: intermediate step to a human mission to Mars



Shielding against GCR Lunar regolith shielding is comparable to or better than a conventional AI structure



Shielding against SPE 50 cm regolith equivalent to the shielding effectiveness of 50 cm SPE vehicle shelter

Susanna Gu

Moon regolith can be considered as a shielding option

Summary

Inflatable Habitat + shielding

- Hadronic interactions are significant, systematics is under control
- The shielding capabilities of an inflatable habitat are comparable to a conventional rigid structure
- Water / polyethylene are equivalent
- Shielding thickness optimisation involves complex physics effects
- An additional shielding layer, enclosing a special shelter zone, is effective against SPE

Moon Habitat

- Regolith shielding limits GCR and SPE exposure effectively
- Its shielding capabilities against GCR can be better than conventional AI structures as in the ISS

Conclusions

- This project represents the first attempt in the European AURORA programme to estimate the radiation protection of astronauts quantitatively
 - Quantitative evaluations
 - Based on open source, validated software
 - Guidance to space industry
- The software system developed is publicly available to the scientific community
 - Advanced software technology
 - Open to extension and evolution

1st development cycle of a long-term project in collaboration with ESA

Thanks to all REMSIM team members for their collaboration, in particular to: V. Guarnieri, C. Lobascio, P. Parodi and R. Rampini

This project was funded by the European Space Agency

Susanna Guatelli